Lecture 15 Block Diagrams of State Space Representations (2/26/20)

Example **1** [(7.9) on p.469 & EXAMPLE 7.11 on p.481.]

Let $G(s) = \frac{s+2}{s^2+7s+12}$ with the sequence of block diagram representations in Figure 1.





$$s^{2}X(s) = -7sX(s) - 12X(s) + U(s)$$
 and $Y(s) = sX(s) + 2X(s)$.

Let $X_1(s) \stackrel{\Delta}{=} sX(s)$ and $X_2(s) \stackrel{\Delta}{=} X(s)$. Then these equations become:

$$sX_1(s) = -7X_1(s) - 12X_2(s) + U(s)$$
 and $Y(s) = X_1(s) + 2X_2(s)$.

Remark 1. The bottom diagram in Figure 1, while more complicated than G(s), provides a number of advantages:

(A1) Clear direction of actually implementing it in hardware that includes integrators, multipliers, and summing junctions. Op-amps can perform all three functions. The diagram shows that the transfer function denominator terms are fed back, while the numerator terms are fed forward.

(A2) Amenability to multi-input/multi-output (MIMO) high-dimensional settings associated with many real world settings. □

The equivalent time-domain equation are:

 $\dot{x}_1(t) = -7x_1(t) - 12x_2(t) + u(t)$ and $y(t) = x_1(t) + 2x_2(t)$.

These, in turn, lead to the *state space* equations:

$$\dot{\mathbf{x}} = \begin{bmatrix} \dot{x}_1 \\ \dot{x}_2 \end{bmatrix} = \begin{bmatrix} -7 & -12 \\ 1 & 0 \end{bmatrix} \begin{bmatrix} x_1 \\ x_2 \end{bmatrix} + \begin{bmatrix} 1 \\ 0 \end{bmatrix} u = \mathbf{A}\mathbf{x} + \mathbf{B}u \quad \text{and} \quad y = \begin{bmatrix} 1 & 2 \end{bmatrix} \mathbf{x} + 0u = \mathbf{C}x + \mathbf{D}u.$$

Matlab:

 $> a=[1 7 12]; > b=[1 2]; > G=tf(b,a) = (s+2)/(s^2+7s+12)$

> [A,B,C,D]=tf2ss(b,a); A = [-7 -12; 1 0]; B = [1; 0]; C = [1 2]; D = 0.

Conclusion: The tf2ss command computes the *controller canonical form* of the state.

The *controllability matrix* is: $C = \begin{bmatrix} \mathbf{B} & \mathbf{AB} & \cdots & \mathbf{A}^{n-1}\mathbf{B} \end{bmatrix}$. The matrices (**A**,**B**) will be *controllable* if and only if *C* is nonsingular. The authors note that "controllability is a function of the *state* of the system, and cannot be decided from a transfer function." They then state (p.477) that "To discuss controllability more at this point would take us too far afield." I agree.

Now let's arrive at yet another representation.

$$\ddot{y} + 7\dot{y} + 12y = \dot{u} + 2u$$
 gives: $\dot{y} = -7y + u + \int (-12y + 2u)dt$.

Let
$$x_1 = y$$
 and let $x_2 = \int (-12y + 2u) dt$. We then have: $\dot{x}_1 = -7x_1 + x_2 + u$. (1)

Now, $x_2 = \int (-12y + 2u) dt$ gives: $\dot{x}_2 = -12x_1 + 2u$. (2)

From (1) and (2):
$$\dot{\mathbf{x}} = \begin{bmatrix} \dot{x}_1 \\ \dot{x}_2 \end{bmatrix} = \begin{bmatrix} -7 & 1 \\ -12 & 0 \end{bmatrix} \begin{bmatrix} x_1 \\ x_2 \end{bmatrix} + \begin{bmatrix} 1 \\ 2 \end{bmatrix} u = \mathbf{A}_0 \mathbf{x} + \mathbf{B}_0 u \text{ and } y = \begin{bmatrix} 1 & 0 \end{bmatrix} \mathbf{x} + 0u = \mathbf{C}_0 x + \mathbf{D}_0 u.$$

This is called the *observer canonical form*. The block diagram is Figure 7.10 on p.476.

In this case the controllability matrix is: $C_0 = \begin{bmatrix} \mathbf{B}_0 & \mathbf{A}_0 \mathbf{B} \end{bmatrix}$. This matrix will be singular if the transfer function zero at -2 were to drift to either -3 or -4. This is because G(s) would then experience a pole/zero cancellation.

Generalization of the controller canonical state space representation:

For the single-input/single-output (SISO) transfer function

$$G(s) = \frac{b_m s^m + \dots + b_1 s + b_0}{s^n + \dots + a_1 s + a_0} \quad ; \quad m \le n ,$$

the *controller* state space representation for this single-input/single-output system is:

$$\dot{\mathbf{x}} = \begin{bmatrix} \dot{x}_1 \\ \dot{x}_2 \\ \vdots \\ \dot{x}_n \end{bmatrix} = \begin{bmatrix} -a_{n-1} & -a_{n-2} & \cdots & -a_0 \\ 1 & 0 & \cdots & 0 \\ 0 & 1 & \cdots & 0 \\ 0 & \cdots & 1 & 0 \end{bmatrix} \begin{bmatrix} x_1 \\ x_2 \\ \vdots \\ x_n \end{bmatrix} + \begin{bmatrix} 1 \\ 0 \\ \vdots \\ 0 \end{bmatrix} u = \mathbf{A}\mathbf{x} + \mathbf{B}u \quad \text{and} \quad y = \begin{bmatrix} b_m & b_{m-1} & \cdots & b_0 \end{bmatrix} \mathbf{x} + 0u = \mathbf{C}x + \mathbf{D}u$$

[Note: These equations are (7.13) on p.470 in the book.]

Remark 2. There are *many* other state space representations for a given system. The book covers them in detail. In this course we will not pursue them in such detail. Nonetheless, it is important to be aware that a given system can be represented in different ways.

Now, if we have time, let's have some fun ©



It can be shown that the -3dB BW for a pulse of width Δ is: $\omega_{BW} \approx 2.78 / \Delta = 2.78 / 2 = 1.39 rad / s$.

We also have: >> eigs(A) = [-8.4322 -0.4845 +/-2.3329i -0.0088]. Hence, the Dutch roll damped natural frequency is 2.33 rad/sec. This is above the input -3dB BW frequency 1.39 rad/sec. Hence, this mode will not be as strongly excited as it would be, were the rudder pulse to have a shorter but greater shape.

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 10^{3}

After that response in $\beta(t)$ has decayed, what remains is very low frequency in nature (plot 1). It's early amplitude is ~0.02. The amplitude of r(t) in this region is ~0.12. The ratio 0.02/0.12=1/6 = -15.6 dB; which is what is predicted in the Bode FRF plot. \Box